

# DARPA 10-Year Lunar Architecture (LunA-10) Capability Study: Lunar Rail Network Infrastructure Study

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## LSIC Overview

The views, opinions, and/or findings expressed are those of the author(s) and should not be interpreted as representing the official views or policies of the Department of Defense or the U.S. Government.

This research was developed with funding from the Defense Advanced Research Projects Agency (DARPA).

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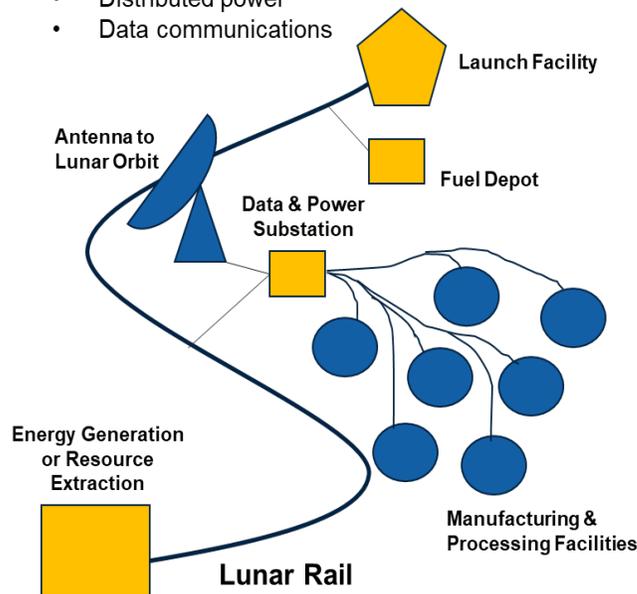
25 April 2024

# Lunar Rail Network: Project Introduction

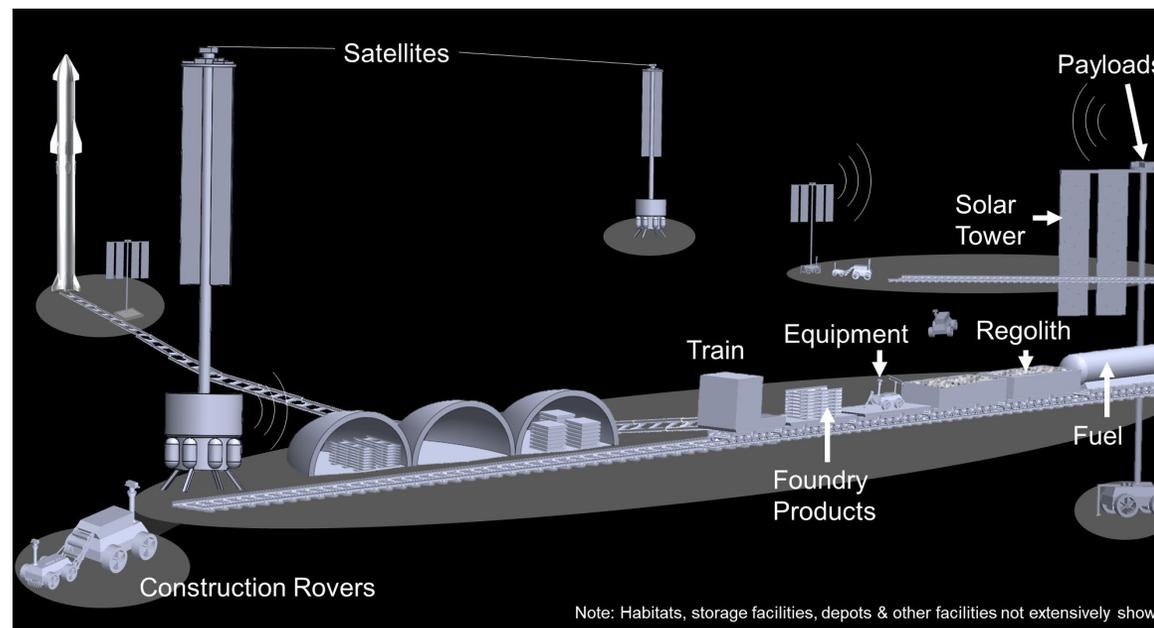
- **Challenges with Rovers:**
  - Dust, Range, Payload Capacity, Speed, Rolling Resistance, Surface Wear, Recurring Cost
- **Potential Solution: Could we build a rail network on the Moon? Should We?**
  - Move large amounts of mass easily and efficiently with minimal impact to surface

**Launch Site Needs:**

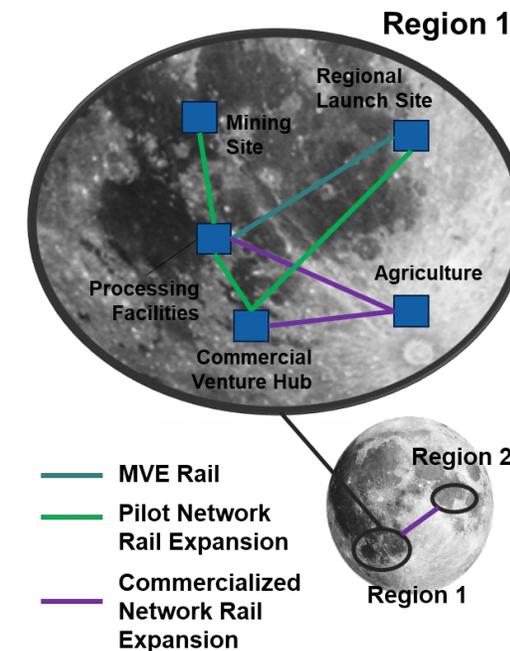
- Propellant delivered to storage depot
- Distributed power
- Data communications



Source: Northrop Grumman Lunar Rail Network Study



Source: Northrop Grumman Lunar Rail Network Study



Notes: Destinations are for illustration only; Not to scale.

Source: Northrop Grumman Lunar Rail Network Study

# LunA-10 & Lunar Rail Mission Space

## Mission A: Surface Transport of Lunar-Derived Materials for Cislunar and Earth Orbit ISAM

[1\) Cost for Space Launch to Low Earth Orbit- Aerospace Security Project \(csis.org\)](#)

[2\) Commercial Lunar Propellant Architecture: A Collaborative Study of Lunar Propellant Production \(isruinfo.com\)](#)

### Projected Value:

(Assuming Starship lowers price of \$1500/kg to LEO<sup>1</sup> to \$300-\$600/kg)

- ~\$1,400-\$2,600/kg to GEO (4x LEO<sup>2</sup>)
- ~\$2,000-\$3,800/kg to Moon (>6x LEO<sup>2</sup>)

### Demand:

- Fuel for Deep Space: 100MT/trip
- Refueling in GEO: 300MT+
- Projected Construction: 100MT's

### Projected Annual Market Value:

- 500MT-2500MT = \$1.6B-\$8B

## Mission C1: Lunar Surface Tourism

**Demand:** Driven by price

**Price:** Per one study, ~\$75M per ticket<sup>3</sup> for a tour that includes surface mobility and 20 passengers (price may come down, and buyers would grow)

### Projected Annual Market Value:

- \$0B to \$1.5B

## Mission B: Surface Resource Mobility for Lunar Surface Scientific Missions

**Missions:** Lunar Telescopes, Solar Weather Monitoring, Biological Understanding, etc...

**Demand:** Multiple per decade (10's to <200)

**Mission Value:** Cost avoidance of bringing power, comm, rovers and equipment. ~10,000 kg per mission = \$38M launch cost plus \$100M+for the equipment → \$150M per mission

### Projected Annual Market Value:

- \$0.5B-\$3B

## Mission C2: Goods from/to Earth

### Potential Demand/Value:

- Moon Souvenirs/Burials (\$20M)<sup>3</sup>
- Other: <\$20M

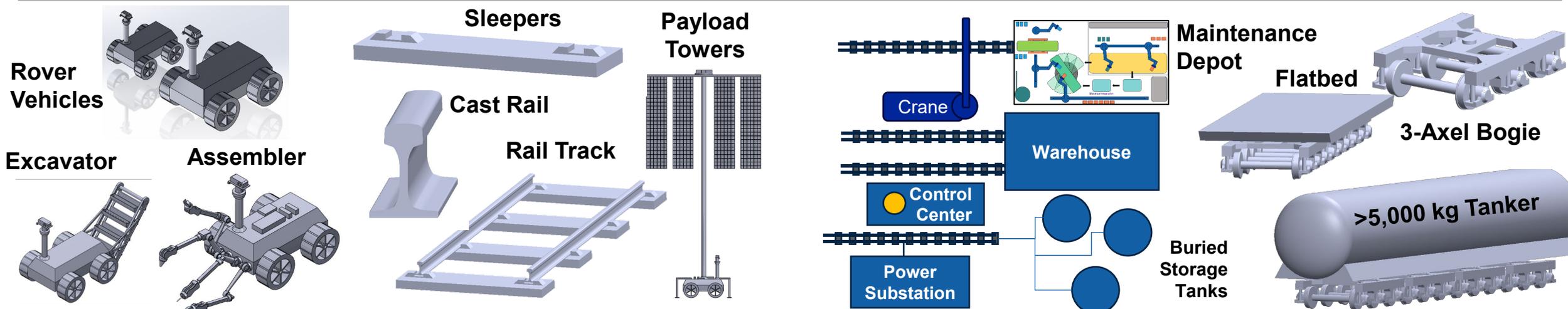
### Projected Annual Market Value

- ~\$0.05B

[3\) Demand Drivers of the Lunar and Cislunar Economy \(newspaceconomy.ca\)](#)

Large mass transport may be key to \$2B to \$13B of annual economic activity by 2035

# Lunar Rail Network Product Overview



## Construction & Maintenance Equipment

- **Construction Rover Vehicles**
  - Med (500kg), Heavy (2500kg)
- **Construction Payloads**
  - Survey, Excavate, Haul, Doze/Compact, Assemble, Supervise, Crane
- **Inspection & Maintenance Payloads**
  - Wear Monitoring, Weld Repair, Component Replace

## Rail Route Infrastructure

- **Embankments**
- **Bridges**
- **Integrated Rail Track**
  - Sleepers, Rails, Fastening System, Switches
- **Additional Service Infrastructure**
  - Comm/Power/PNT Payload Towers
  - Wired Grid
  - Pipelines

## Rail Station Infrastructure

- **Comm/Power/PNT Payload Towers**
- **Command and Control**
  - Control Electronics, Sensors, Data Management/Analytics
- **Traffic & Payload Management**
  - Switches, Loading Docks, Maintenance Depot
- **Resource Distribution Infrastructure**
  - Substations, Data Centers, Storage Tanks, Warehouses

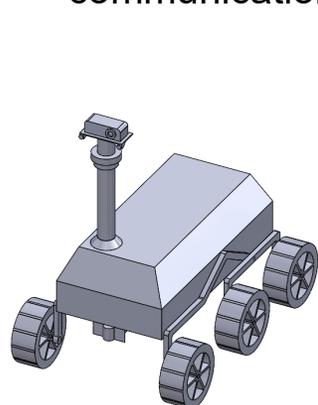
## Integrated Rail Vehicles

- **Rail Vehicle Platform**
  - Couplers, Bogie Attachments, Primary Frame
- **Engine Electronics Bus**
  - Power, Communications, Control, Radiators
- **Payload Transport Assemblies (Tankers, Hoppers, Flatbeds)**
  - Payload Structures, Electrical Control Boxes
- **Bogies**
  - Motors, Brakes, Wheel Sets, Frames, Suspensions

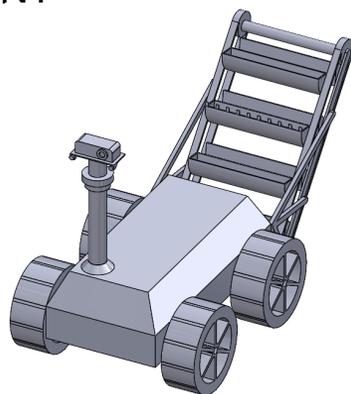
# Foundation Forming Equipment Concepts

- **Surveyor (S):** High-mobility platform with ground penetrating radar and regolith characterization
- **Excavator (E):** Base platform with articulated bucket ladder for excavation with rippers along chain
- **Hauler (H):** Base platform with coverable bin
- **Compactor (C):** Base platform with a vibratory plate compactor, an articulated vibratory dozer for large rocks and coarse/fine grading, and a coverable bin
- **Assembler (A):** Base platform with robotic arms with for placing, sintering, and welding
- **Manager (M):** Base platform with outriggers, solar arrays, and communications/PNT

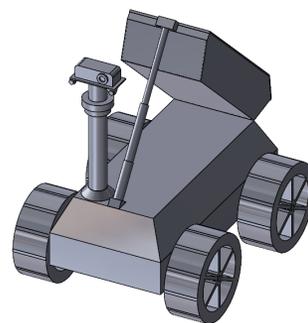
Type	Approx. Duty Cycle	Notional Unit Launch Mass (kg)	Notional Unit Average Power (kW)
Surveyor	90%	500	0.5
Excavator	70%	1,300	2.3
Hauler	70%	1,400	1.0
Compactor	90%	1,500	0.5
Assembler	70%	1,100	2.1
Manager	100%	1,000	0.1



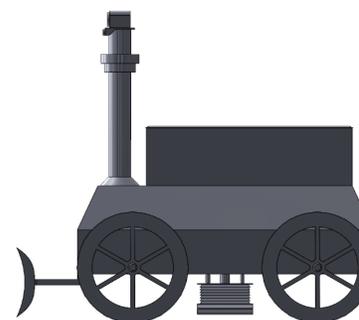
Surveyor



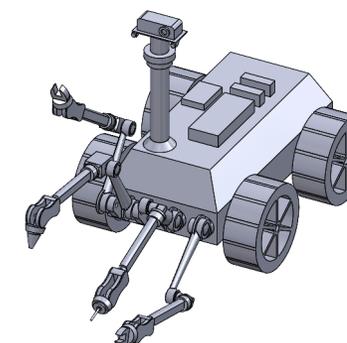
Excavator



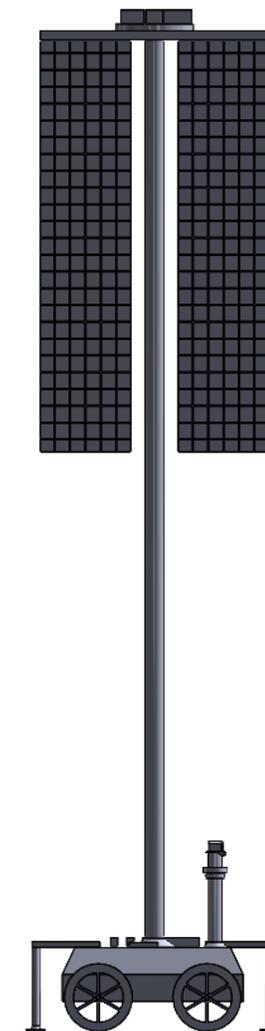
Hauler



Compactor

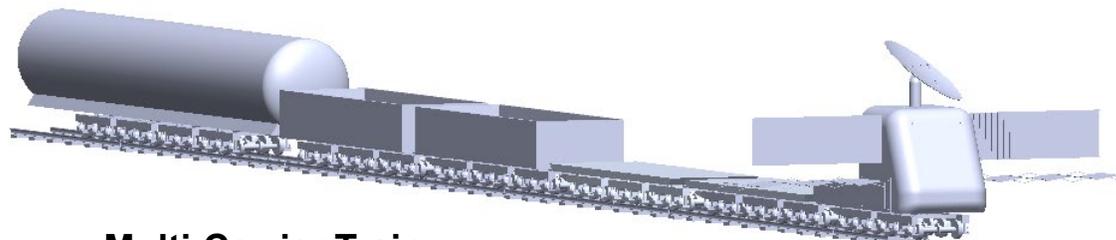


Assembler

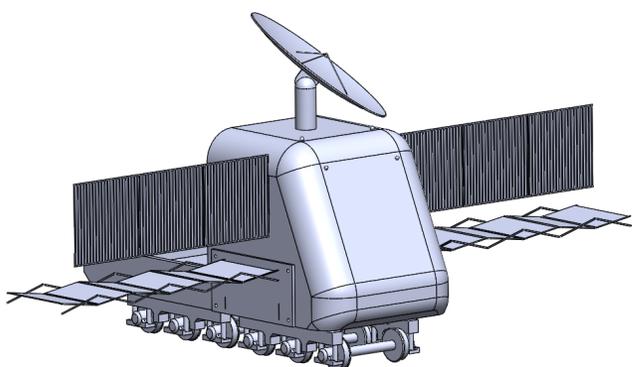


Manager

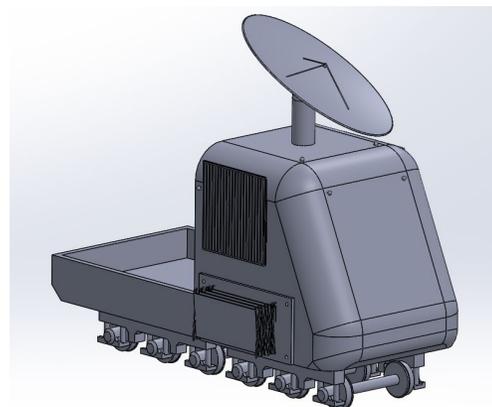
# Rail Network Design & Analysis



**Multi-Carrier Train**



**Lunar Day Configuration**



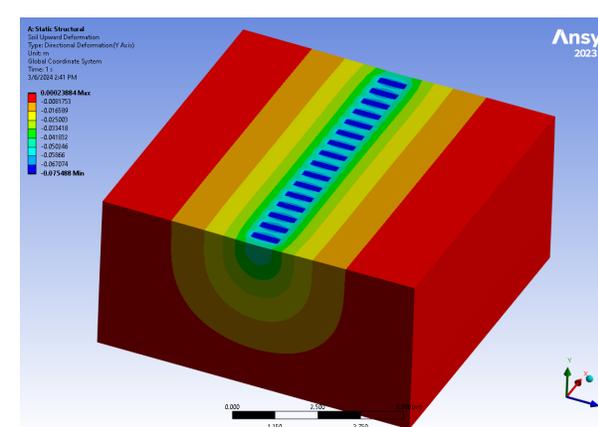
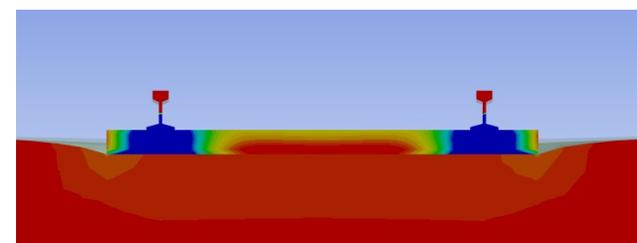
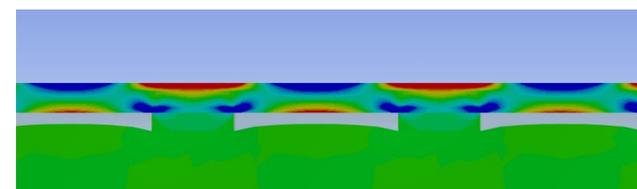
**Lunar Night Configuration**

Train and carrier concept designed with modularity:

- Minimizes launched mass of equipment
- Bogies, frames & transport assembly concepts are reconfigurable to payload transport needs

## Key Design Challenges:

- **Lunar Soil:** Uncertainty in lunar soil modulus of elasticity for compacted regolith → Potential for deflections >5 cm even with well distributed loads
- **Energy:** High energy consumption for sintering regolith or extracting and processing metals → Drives goal to minimize processed material for architecture
- **Gravity:** Reduced gravity vector → reduces traction force and either increases required cant angle in turns or limits speeds in turns

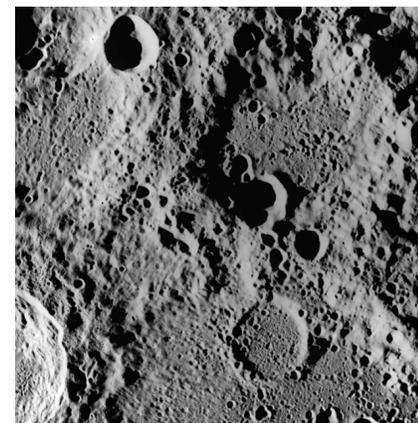


**Lunar Soil Deformation**

# Lunar Rail Materials Challenges

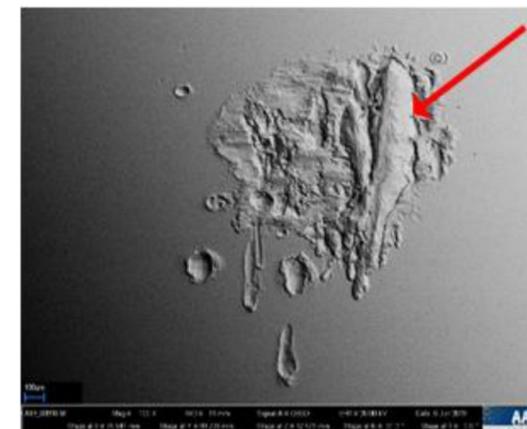
- **Limited In-Situ resource availability**
  - Metal extraction from regolith is challenging
  - Producible alloys differ from common Terrestrial options
- **Extreme temperatures (<-200C to >120C)**
  - Raises concerns for ductile to brittle transitions, thermal expansion, and over-aging
- **Vacuum environment**
  - Lack of surface oxide accelerates wear and cold welding
- **High stress between the wheel and rail**

Lunar surface



Source: [Apollo Image Archive \(asu.edu\)](https://apolloimagearchive.asu.edu)

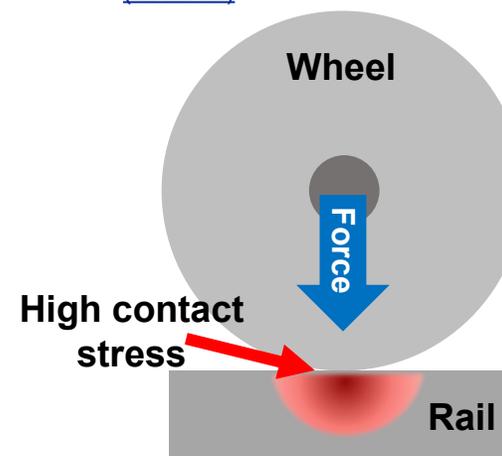
Wear Examples



Source: [Cold Welding in Space Mechanisms Due to Fretting \(mdpi.com\)](https://www.mdpi.com)



Source: [Ductile-to-Brittle Transition and Brittle Fracture Stress of Ultrafine-Grained Low-Carbon Steel - PMC \(nih.gov\)](https://pubmed.ncbi.nlm.nih.gov/)



Source: Northrop Grumman Lunar Rail Network Study

**These challenges significantly constrain material selection options**

# Minimum Viable Experiment (MVE) Phase 1 Objectives

- Goals: Demonstrate commercial ISRU resource extraction framework viability:

Perform Mining,  
ISRU & Foundry  
Operations

Perform Minimum Viable  
Lunar Surface  
Infrastructure Construction

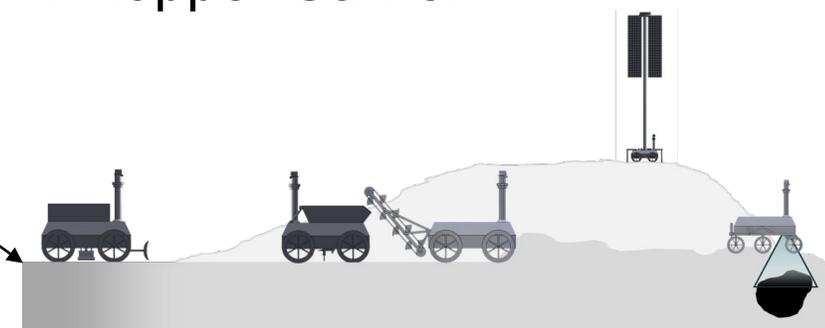
Perform  
Minimum Viable  
Rail Transport

Execute First  
Commercial Sale  
of Lunar Fuel

- MVE Operations Equipment:** 2 Flatbeds (1 Motorized) + 1 Hopper Carrier

- Potential MVE Phase 1 Lunar Rail Activities:**

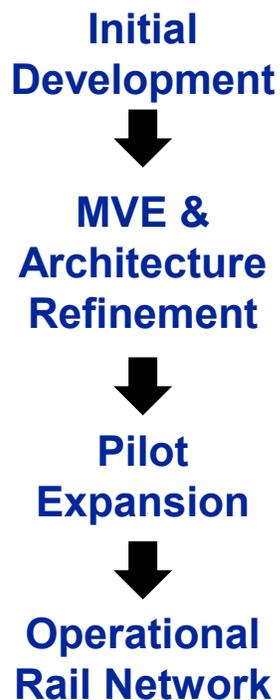
- Site preparation (surveying, excavating & compacting)
- Load hopper with regolith during route construction (demo regolith loading/delivery), deliver regolith to ISRU station & resource processing payloads
- Load engine/flatbed #1 with finished foundry and cast products and deliver to construction fleet (demo goods delivery)
- Use produced rails & sleepers in construction of rail network length
- Install power nodes, demonstrate secondary services
- Place tank on flatbed #2, transport fuel (demo tanker transport) and launch
  - Would bring or harvest a spent propellant tank
  - May need a small propellant tug with launch system to return fuel to orbit



**NOTE:** Performing these activities is not in scope of the LunA-10 study

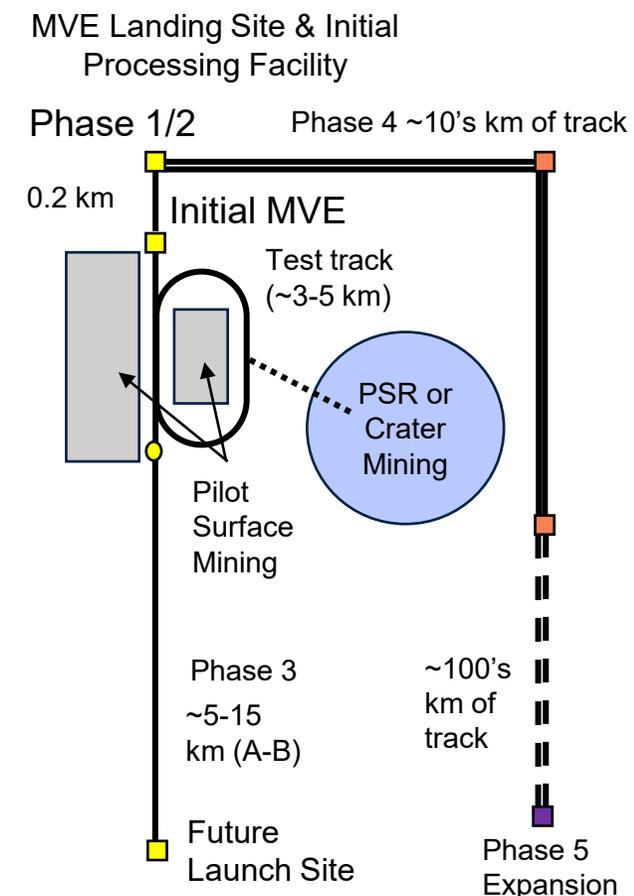
# MVE to Pilot to Operational Expansion

- **Minimum Viable Experiments (MVE):** Prove Lunar surface viability and refines architecture design
- **Pilot Expansion:** Uses second generation carriers/engines & construction equipment (same track gauge)
- **Operational Rail Network:** Settlement 1 infrastructure complete, fully operational and generating revenue



Years	Development Phase	Phase Activities*
2	Scientific Understanding	Gap mitigation & fundamental science
2	MVE Development	Non-Recurring Engineering, Manufacturing, Assembly, Equipment Testing & Demos
1	MVE Integration, Test & Launch/Transit	Integration, test, launch & transit to Lunar surface
1	Phase 1 MVE	Construct rail of ~0.2 km, perform MVE Phase 1 Demo
2	Phase 2 MVE	Add dynamics, civil engineering & wear test track to mature railroad architecture. Support early mining & foundry.
1	Phase 3 Pilot	Expand to connect to high traffic launch site
1	Phase 4 Pilot	Add additional routes to other processing facilities around 1 <sup>st</sup> settlement, scale towards full operational state
3+	Phase 5 Operational Expansion	Expand reach to additional settlements (investments justified for each addition)

## Concept of Expansion

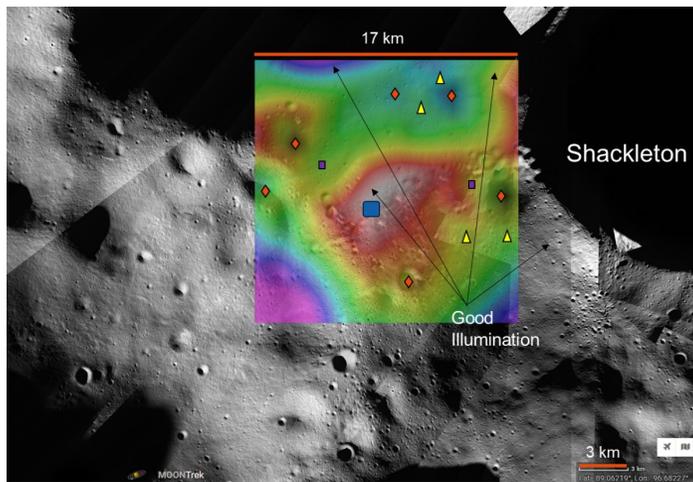


**Lunar Rail Network would Scale as Lunar Surface Utilization Grows**

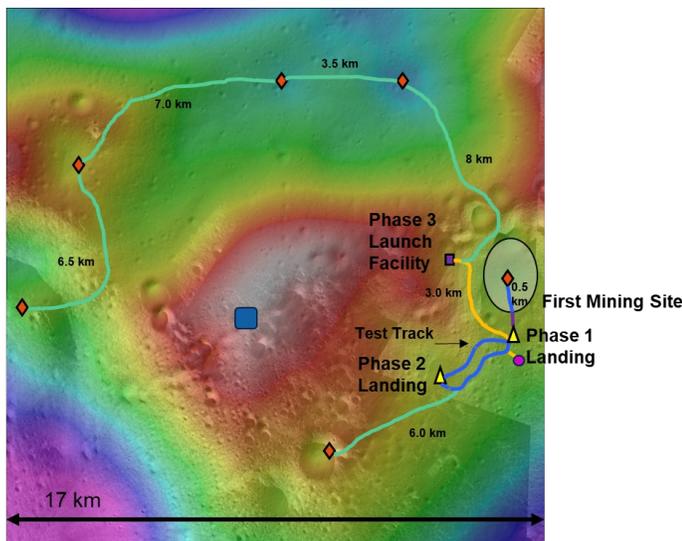
**\*NOTE:** Performing these activities is not in scope of the LunA-10 study

# MVE Site Selection & Expansion

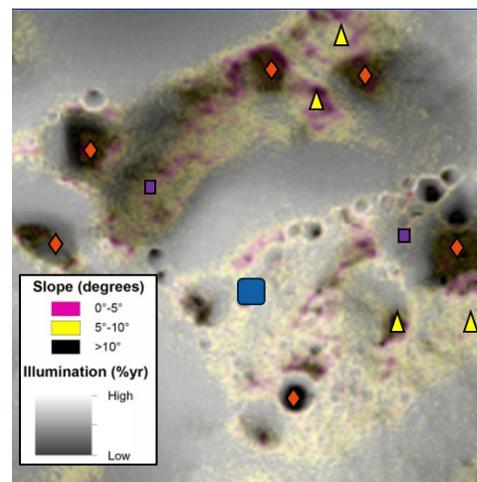
Elevation and Regional Context<sup>1</sup>



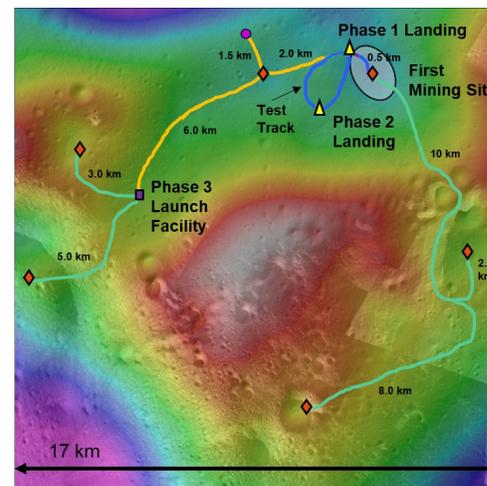
**Lunar Rail Network Study:  
Design Reference Plan 1**



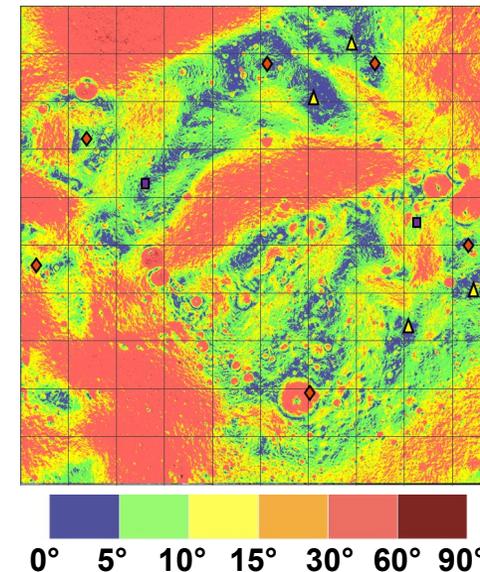
Illumination & Slope<sup>2</sup>



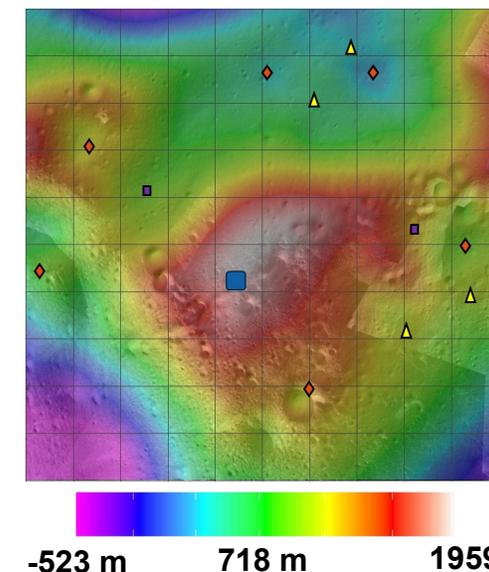
**Lunar Rail Network Study:  
Design Reference Plan 2**



Slope Angle<sup>1</sup>



Elevation<sup>1</sup>



Permanently Shadowed Regions (Mining Resources)	Candidates for Phase 3 Launch Facility
Candidates for Phase 1/2 MVE Landing	Artemis Site 001

<b>Plan 1</b>	<b>Plan 2</b>	<b>Phase 5: ~300-400 km (not shown)</b>
Phase 1: 0.2 km	Phase 1: 0.2 km	
Phase 2: 4.0 km	Phase 2: 4.0 km	
Phase 3: 4.5 km	Phase 3: 9.5 km	
Phase 4: 31 km	Phase 4: 28 km	

Sources:  
(Adapted from)

- 1) Barker, M.K., et al. (2020), Improved LOLA Elevation Maps for South Pole Landing Sites: Error Estimates and Their Impact on Illumination Conditions, Planetary & Space Science, in press, doi:10.1016/j.pss.2020.105119.
- 2) Stopar J. and Meyer H. (2019) Annual Illumination and Topographic Slope of the Moon's South Polar Ridge, Lunar and Planetary Institute Regional Planetary Image Facility, LPI Contribution 2179, <https://repository.hou.usra.edu/handle/20.500.11753/1264>

# Investment & Trip Price by Phase Summary

## Non-Recurring Engineering & Operations

- Concept Development and Design of Infrastructure and Transport System
  - Train, Civil Structures, Substations, Construction Equipment
- Launch of material to Moon Base
- Test track and equipment/system maturation
- Lunar Rail staged infrastructure buildup
- Manufacturing, integration, test and launch of construction equipment and Earth-sourced components/assemblies

## Recurring Engineering & Operations

- Operational cost of train (energy, comms/PNT, loading/unloading)
- ISRU-derived manufacturing of new elements and maintenance of existing elements (Railcars, beds, etc.)

Source: Northrop Grumman Lunar Rail Network Study

Parameter	Phase 1	Phase 2	Phase 3	Phase 4	Phase 5
Average Trip Length [km]	0.2	2.4	3.9	24	128
Average Trip Payload [kg]	200	5,000	32,000	45,000	83,000
Total Launched Mass by Phase [kg]	16,100	25,300	64,200	90,000	167,000
ISRU Metallic Mass Required [kg]	1,800	43,300	212,000	520,000	3,700,000
<b>TOTAL Estimated NRE [\$B]</b>	<b>0.5-1.0</b>	<b>0.5-1.0</b>	<b>2.0-3.0</b>	<b>2.0-3.0</b>	<b>4.0-5.0</b>
<b>TOTAL Estimated RE [\$B]</b>	<b>---</b>	<b>&lt;0.1</b>	<b>0.1-0.2</b>	<b>0.3-0.5</b>	<b>2.0-3.0</b>
<b>TOTAL Goal Avg Price per Trip [\$M]</b>	<b>N/A</b>	<b>8.0</b>	<b>8.0</b>	<b>5.0</b>	<b>2.5</b>
<b>Mass Price Efficiency [USD / kg-km]</b>	<b>17,700,000</b>	<b>660</b>	<b>51</b>	<b>3.4</b>	<b>0.23</b>

**Note:** Numbers shown are to ballpark costs and do not include contingency. At the final system concept review the numbers will be updated along with the assumptions below.

### Pilot Lunar Rail – Key Architecture Assumptions

- Max Transport Mass: 100,000 kg – 125,000 kg
- Cost Assumptions:
  - Development Cost - \$30,000 per kg<sup>1</sup> (Equipment, Detailed Components)
  - Development Cost - \$3,000 per kg (Maintenance Depot Building)
  - Cost of Launch to Moon Base - \$10,000 per kg (~7x Falcon 9 to LEO)<sup>2</sup>
  - Cost of Manufacturing Metal Parts on the Moon - \$500 per kg
  - Cost of Manufacturing Equipment on Earth - \$5,000 per kg
  - Cost of Energy on the Moon - \$100 per kWh
  - Maintenance: 5% per year of manufacturing cost for infrastructure (20-year life), 20% per year for carriers (5-year life)

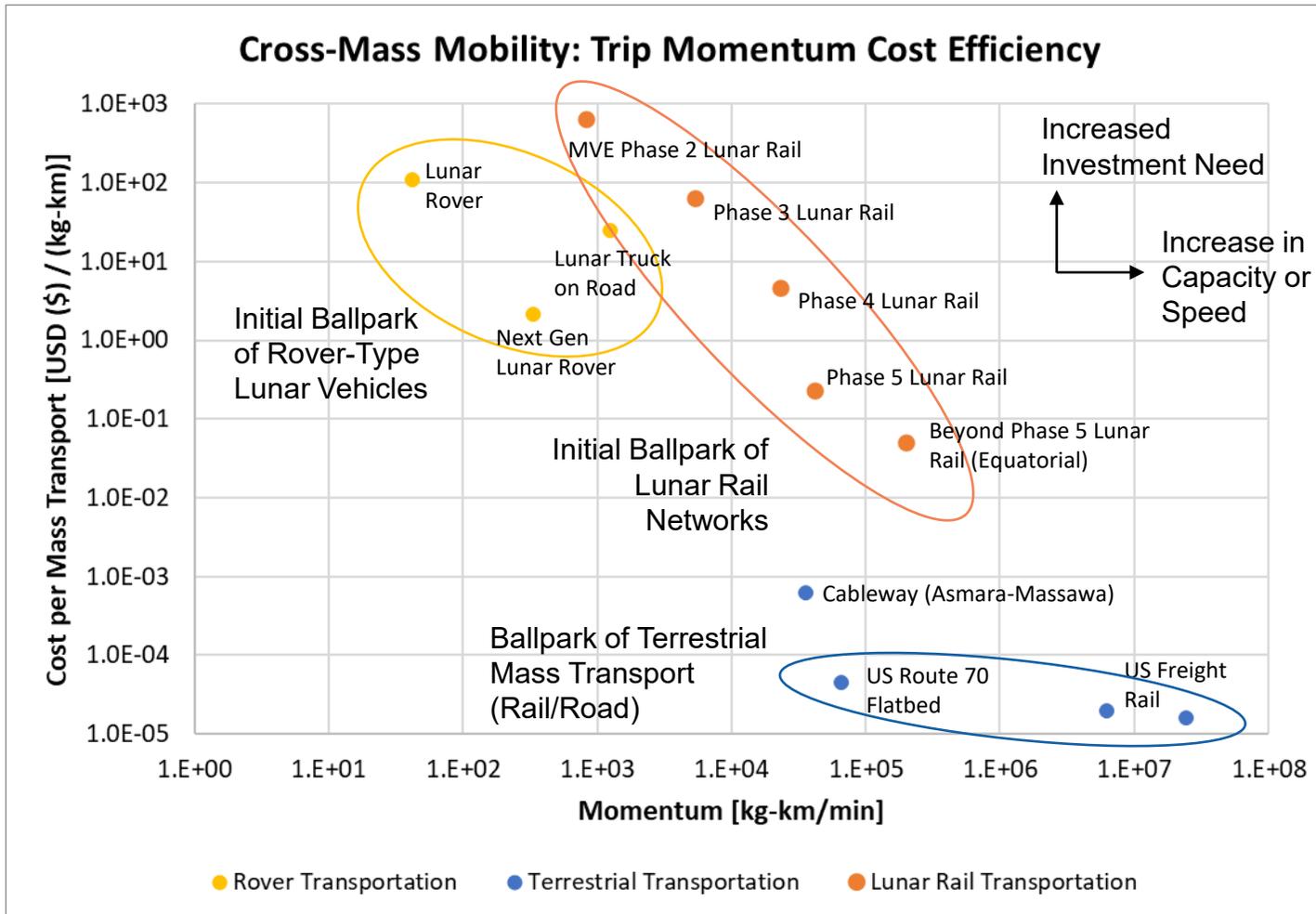
**Phase 2+ Price Per Trip: ~\$2.5M-\$8M**

Additional Sources: 1) [Demand Drivers of the Lunar and Cislunar Economy \(newspacereconomy.ca\)](https://www.newspacereconomy.com), Page 123  
 2) [Cost for Space Launch to Low Earth Orbit- Aerospace Security Project \(csis.org\)](https://www.csis.org)

# Cross-Mass Mobility: Trip Momentum Cost Efficiency

## White Space Plot Draft Result

**Note:** Derivation of the terrestrial transportation and rover values will be discussed in the LunA-10 Final Report



- **Initial Take-Aways:**
  - Rover constrained by capacity
  - Lunar rail constrained by infrastructure cost
  - Slope is steeper with higher infrastructure cost, offset by increasing capacity of transport
  - Refinement of total costs and capacities is important to understanding rover vehicle versus rail overlap region
  - Developing projected transport service demand is important to determine when rail makes strict economic sense with all other things being equal

Cross-Mass Mobility: Momentum Cost Efficiency		
Transport Methodology	X Axis: kg*km/min	Y-Axis: \$/kg-km
Lunar Rover	42	109
Lunar Truck on Road	1250	24.7
Next Gen Lunar Rover	333	2.13
Phase 2 Lunar Rail Transport	830	656
Phase 3 Lunar Rail Transport	5400	63.6
Phase 4 Lunar Rail Transport	23000	4.65
Phase 5 Lunar Rail Transport	42000	0.23
Beyond Phase 5 Lunar Rail Network	200000	0.05
Cableway (Asmara-Massawa)	35900	0.000612
US Route 70 Flatbed	66000	0.000045
US Freight Rail Transport (Long)	24500000	0.000016
US Freight Rail Transport (Short)	6250000	0.000020

There is a rover to rail cross-over point around >1000 kg-km/min and/or achieving <\$5-10/kg-km

**NORTHROP**  
**GRUMMAN**

The logo graphic consists of a thick black horizontal line extending from the right side of the word "NORTHROP" to the right edge of the frame. From the right end of this horizontal line, a thick black vertical line extends downwards to the right edge of the frame, forming an L-shaped corner.