

A large, semi-transparent globe is positioned on the left side of the slide. A bright, purple, fan-like light beam originates from behind the globe and extends towards the center of the slide.

F6 PIVOT Pleiades Innovative VCDM Optimization Tool

February 06, 2009

Innovation You Can Count On™

ORB
Listed
NYSE
THE NEW YORK STOCK EXCHANGE



PIVOT Overview



- **Lifecycle Cost, Value, and Risk Assessment Tool**
 - Monte-Carlo simulation with easy to modify parameter selections/levers
- **Coded in MATLAB**
 - Preferred platform for interaction with Georgia Tech.
 - Future considerations for integration of other tools (example STK)
 - **Widespread Use**
 - No additional installation steps (if MATLAB already installed)
 - Simple instructions to execute scripts
- **Incorporates uncertainty models to evaluate design performance**
 - Launch / On-Orbit Failures, Schedule Slip
- **Integrated with Design Sizer to rapidly evaluate design tradespace**
 - **Discrete relations**
 - Based on conceptual designs
 - **Parametric relations (rubberized)**
 - From Georgia Tech's GT-FAST design sizer



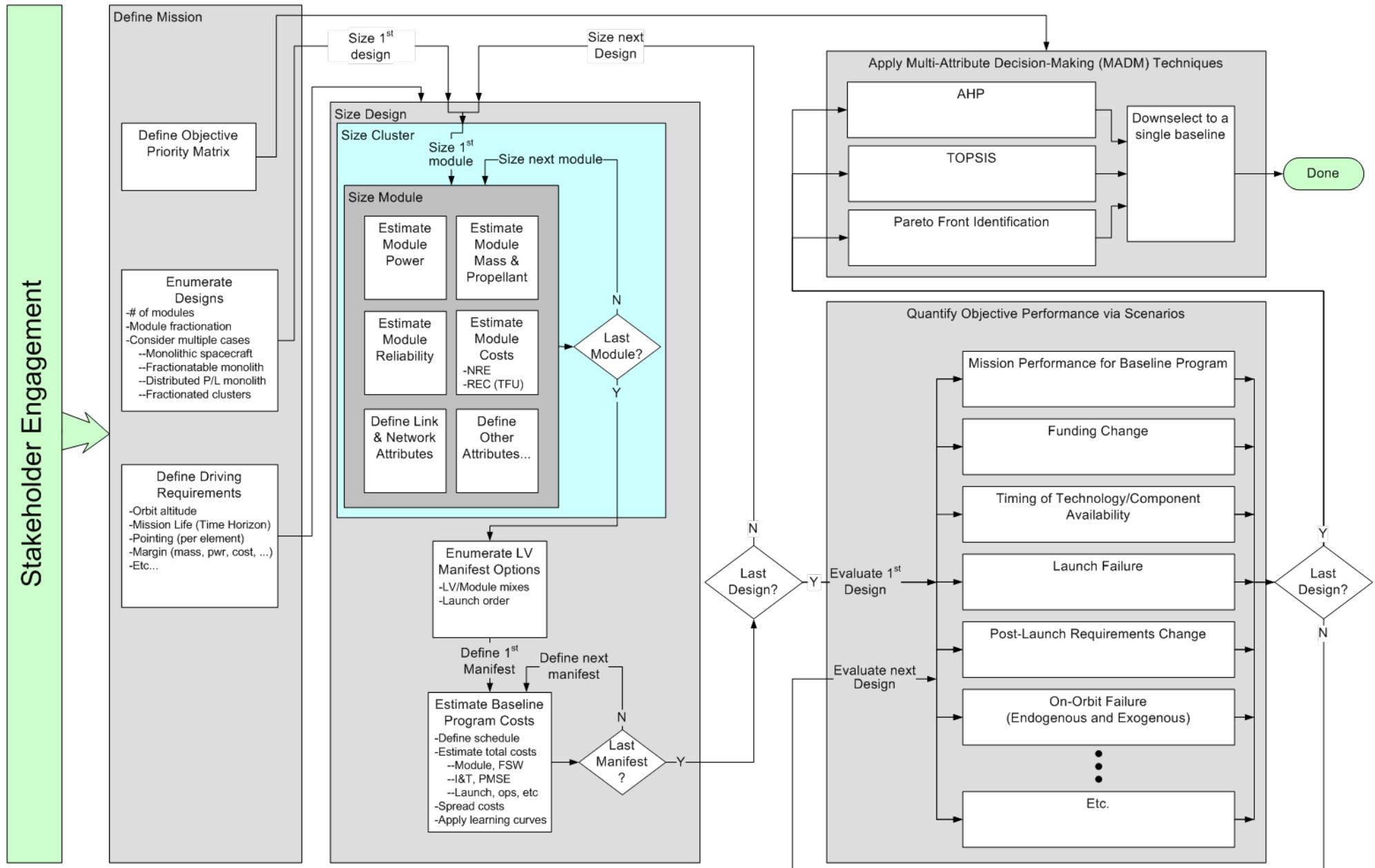
IBM JPL MIT

Massachusetts
Institute of
Technology

Georgia Tech



General Flow





Design Sizer – Design Parameters



○ Discrete Evaluations

- **Mass Budget**
 - Includes Delta V Budget

- **Power Budget**
- **Reliability (FIT values)**
- **Cost - NRE and RE**

○ Parametric (non-proprietary)

- **From Georgia Tech - GTFAST**
- **Mass**
- **Power**
- **Reliability**
 - Uses FIT values of discrete
- **Cost**
 - SSCM (from GTFAST)
 - SMAD for public releasable version



IBM JPL MIT

Massachusetts
Institute of
Technology

Georgia Tech



Value Model Flow

- **Module phase evaluation**
 - Determines design and launch schedule of each module
 - Evaluates current mission phase of each module (Development, Launch, ICO, Ops)
 - Applies cost based on applicable phase
 - **Models**
 - Cost Uncertainty Model
 - TRL Schedule Change Model
 - Launch Failure Model
 - Insurance Model
 - Funding Change Model (stand alone version - to be improved/integrated in Phase 2)
- **Reliability and revenue models**
 - Failure assessment of active components
 - Incorporates link budget analysis to determine preferred DL paths
 - Evaluates revenue using pricing models for each mission
 - **Models**
 - On-Orbit Failure Model
 - Value / pricing models
- **Replenishment decision**
 - Determines if and when modules will be replenished
 - **Models / Selections**
 - Replenishment
 - LV Replenishment Strategy
 - Long-Lead Item
- **Value Assessment**
 - NPV, cumulative cost and revenue computations





Value Assessment Model Inputs



- **Module Configuration**
 - Generated from design sizer or point design
 - Data Structure of cost and performance parameters
- **Launch Vehicle Configuration**
 - Generated from design sizer or point design
 - Data Structure of cost and performance parameters
- **Model Selections**
 - Simulation parameters (MonteCarlo runs, design horizon)
 - Uncertainty Models
- **Model Parameters**
 - Cost Uncertainty Parameters
 - Value Levers
- **Default values defined in pub_Model_Enable_Selection_default.m**



IBM JPL MIT

Massachusetts
Institute of
Technology

Georgia Tech



Model Parameters



- Cost Uncertainty Parameters

- Module Cost
- Launch Vehicle Cost
- Operations Cost

- Cost / Value Levers

- Discount rate
- Module learning curve rate
- LV learning curve rate
- Insurance cost factor
- Inflation rate
- Operations cost (per module)
- Link cost rates
- Long-lead item time

- Default values defined in pub_Model_Cost_Parameters_default.m



IBM JPL MIT

Massachusetts
Institute of
Technology

Georgia Tech



Value Assessment Model Data Structures



○ Module Configuration Input

- ‘module_config_default’ in pub_scenario_main.m
- Generated from design sizer or point design
- Data structure with the following parameters
 - row # identifies module #
 - Column 1 - Wet Mass
 - Column 2 - NRE
 - Column 3 - RE
 - Column 4 - module_life
 - Column 5 - Design/I&T time
 - Column 6 - ICO time
 - Column 7 - TRL
 - Column 8 - LV assigned
 - Column 9 - F6 Comp array

○ Launch Vehicle Configuration Input

- ‘lv_config_default’ in pub_scenario_main.m
- Generated from design sizer or point design
- Data structure with the following parameters
 - Row # identifies LV #
 - Column 1 - configuration of which modules on each LV
 - Column 2 - reference for # of modules launched on LV
 - Column 3 - Launch Vehicle Type
 - Column 4 - Launch Vehicle Reliability
 - Column 5 - Launch Vehicle Cost
 - Column 6 - Launch Time (Months)



IBM JPL

Mit
Massachusetts
Institute of
Technology

Georgia Tech



Value Assessment Model Data Structures



○ Module Summary Table

- **Lifecycle summary of all modules**
 - Includes baseline configuration and replenishments
- **'module_summary' in pub_scenario_main.m**
- **Data structure with the following parameters**
 - Column 1 - Module counter (for production learning curve)
 - Column 2 - Module ID #
 - Column 3 - Design start time
 - Column 4 - Module Cost
 - Column 5 - LV Cost
 - Column 6 - Module TRL
 - Column 7 - Insurance Cost (if insurance model enabled)
 - Column 8 - Time of Failure
 - Column 9 - On-orbit (operations) time



IBM JPL MIT

Massachusetts
Institute of
Technology

Georgia Tech



Value Assessment Model Data Structures



○ Launch Summary Table

- Lifecycle summary of all launches
- ‘lv_summary’ in pub_scenario_main.m
- Data structure with the following parameters
 - Row # indicates launch attempt #
 - Column 1 - Configuration of which modules aboard
 - Column 2 - reference for # of modules launched
 - Column 3 - LV cost
 - Column 4 - Time of launch
 - Column 5 – Success (1) or Failure (-1)



IBM JPL MIT

Massachusetts
Institute of
Technology

Georgia Tech



Value Assessment Model Data Structures



○ Component State Array

- Tracking summary of component states for all modules
- ‘Data structure with the following parameters
 - Row # indicates module #
 - Column 1 - low rate space-ground link
 - Column 2 - high rate space-space link
 - Column 3 - high rate space-ground link
 - Column 4 – low rate 24/7 link
 - Column 5 – Core module
 - Column 6 – SSR (maps to fractionated component 4)
 - Column 7 – MDP (maps to fractionated component 5)
 - Column 8 – Payload 3 (on all modules)
 - Column 9 – Payload 1 (maps to fractionated component 2)
 - Column 10 – Payload 2 (maps to fractionated component 6)



IBM JPL MIT

Massachusetts
Institute of
Technology

Georgia Tech



Value Assessment Model Data Structures



○ Module Statistics Tracking Table

- Tracking summary of modules through phases
- ‘module_tracking_table’ in pub_scenario_main.m
- Data structure with the following parameters
 - Row # indicates module #
 - Column 1 - Module design start (month)
 - Column 2 - Design phase flag (1 if module in design)
 - Column 3 - Design percent complete counter
 - Column 4 - Scheduled launch time
 - Column 5 - Launch phase flag (1 if module ready to launch)
 - Column 6 - ICO phase flag (1 if module in ICO)
 - Column 7 - ICO percent complete counter
 - Column 8 - Scheduled Ops start time
 - Column 9 - Ops phase flag (1 if module in Ops)
 - Column 10 - Ops percent complete counter
 - Column 11 - Module design delay start time (TRL Schedule slip model)
 - Column 12 - Module design delay length (TRL Schedule slip model)
 - Column 13 - Module decision to replenish flag (set when failure or retired)
 - Column 14 - Module to be replenished flag (set when module will be replenished)
 - Column 15 - Module previous launch (currently spare)
 - Column 16 - Module counter reference



IBM JPL MIT

Massachusetts
Institute of
Technology

Georgia Tech



Value Assessment Model Data Structures



○ Monte-Carlo Summary Table

- Comprehensive Output of value model
- Can be used to recreate specific test run
- ‘MC_Summary’ in pub_scenario_main.m
- Data structure with the following parameters
 - Row # indicates test Case #
 - Column 1 - Total Program Cost
 - Column 2 - NPV
 - Column 3 - module_tracking_table
 - Column 4 - module_summary
 - Column 5 - launch_summary
 - Column 6 - Module lifecycle cost (module cost per month – each module)
 - Column 7 - Module lifecycle schedule (module phase per month – each module)
 - Column 8 - Module lifecycle revenue (module revenue per month – each module)
 - Column 9 - Cumulative NPV (by month)
 - Column 10 - Cumulative Program Cost (by month)
 - Column 11 - Cumulative Program Revenue (by month)
 - Column 12 - PL2 Cumulative NPV (by month)
 - Column 13 - PL 3 Cumulative NPV (by month)
 - Column 14 - PL1 Cumulative NPV (by month)
 - Column 15 - Total high rate data per month
 - Column 16 - Total low rate data per month
 - Column 17 - Total 24/7 data per month
 - Column 18 - Pipe Capacity summary per month
 - Column 19 - Program Cost per month



IBM JPL MIT

Massachusetts
Institute of
Technology

Georgia Tech



Value Assessment Model Data Structures



○ Component State Table

- State of modeled components (1 - active, 0 - not on module or failed)
- Matrix of 'N' modules by 10 components
 - Next row populated when module reaches orbit
 - Updated every timestep (On-orbit failures or module retirement)
- 'Data structure with the following parameters'
 - Row # indicates module #
 - Column 1 - low rate DL
 - Column 2 - mission space-space
 - Column 3 - high rate DL
 - Column 4 - 24/7
 - Column 5 - core module
 - Column 6 - solid state data recorder
 - Column 7 - mission data processor
 - Column 8 - payload 1
 - Column 9 - payload 2
 - Column 10 - payload 3



IBM JPL MIT

Massachusetts
Institute of
Technology

Georgia Tech



Sample Cases General Format



- Run File – sample_test_case.m
- 1) Set global parameters
 - Set orbit parameters, module lifetime
 - Set design global parameters
 - pub_globals_bus_types.m
 - Defines globals related to the core spacecraft bus (mass, power, cost, FIT, TRL)
 - Allows parameters to be modified
 - Future work → to be driven by GUI
 - pub_globals_f6_elements.m
 - Defines fractionation elements (mass, power, cost, FIT, TRL) & invalid combinations
 - Future work → to be driven by GUIs
 - Load value global parameters
 - pub_value_globals.m
- 2a) Define scenario / perturbation model states
 - Simulation settings and state of uncertainty models
 - pub_Model_Cost_Parameters_default.m
- 2b) Define scenario / perturbation model parameters
 - Cost and value Levers
 - pub_Model_Enable_Selection_default.m
- 3) Select Configuration
 - Module and LV design
 - pub_baseline_config.m
- 4) Run Model
 - pub_scenario_main.m
- 5) Save / Plot Data
 - plot_cost_revenue_pv.m



IBM JPL MIT

Massachusetts
Institute of
Technology

Georgia Tech



Sample Cases Design Tradespace Evaluation



- Run file – pub_F6_VCDM_Input_Script.m
- 1) Set global parameters
 - Set orbit parameters, module lifetime
 - Set design global parameters
 - pub_globals_bus_types.m
 - pub_globals_f6_elements.m
- 2) Set model parameters and load designs
 - Load all fractionation design concepts (load files for up to 10 modules provided)
 - Based on total number of fractionatable elements (set by pub_globals_f6_elements.m)
 - Eliminate Invalid designs (component placement constraints)
 - pub_component_placement_constraints_new.m
 - Load Value global parameters
 - pub_value_globals.m
 - Define Scenario / Perturbation model inputs
 - pub_Model_Cost_Parameters_default.m
 - pub_Model_Enable_Selection_default.m



IBM JPL MIT

Massachusetts
Institute of
Technology

Georgia Tech



Sample Cases Design Tradespace Evaluation



- Continued from – pub_F6_VCDM_Input_Script.m
- 3) Select Configuration
 - Design Sizer
 - Discrete
 - Uses global parameter inputs
 - Parametric
 - Based on Georgia Tech's GTFAST
 - Cost models provided based on SSCM and SMAD (SSCM not for public release)
 - gtfast_calc_cost_pub_SMAD.m
 - Run additional constraints based on sized designs
 - LV manifest
 - Sets Launch combinations to be considered for each launch vehicle
 - Launch vehicle tradespace set in pub_lv_input.m
 - Several manifest strategies to use – based on desired number of combinations to consider
 - pub_lv_manifest.m
 - pub_lv_manifest_limited.m (preferred configurations)
- 4) Run Model
 - Each Design runs through model
 - pub_scenario_main.m
- 5) Save / Plot Data
 - plot_all_combos.m



IBM JPL MIT

Massachusetts
Institute of
Technology

Georgia Tech



Sample Cases

LV Trade Study – Single and Dual Launches



- Run File – pub_test_lv_trade_revenue.m
- 1) Set global parameters
 - Set orbit parameters, module lifetime
 - Set design global parameters
 - pub_globals_bus_types.m
 - pub_globals_f6_elements.m
 - Load value global parameters
 - pub_value_globals.m
- 2a) Define scenario / perturbation model states
 - Simulation settings and state of uncertainty models
 - pub_Model_Cost_Parameters_default.m
- 2b) Define scenario / perturbation model parameters
 - Cost and value Levers
 - pub_Model_Enable_Selection_default.m
- 3) Select Configuration
 - Module design
 - Load baseline design pub_baseline_config.m
 - Overwrite necessary parameters for single launches
 - LV design
 - Currently Hard-Coded for Single and Dual Launch trades
 - Launch vehicle tradespace set in pub_lv_input.m
- 4) Run Model
 - pub_scenario_main.m
- 5) Save / Plot Data
 - Each Simulation added to same figure
 - For separate single and dual launch plots, remove % from figure()
 - lv_trade_ellipse_plot_revenue_pv.m



IBM JPL MIT

Massachusetts
Institute of
Technology

Georgia Tech



Sample Cases Additional High Rate DL Module



- Run File – trade_study_add_HRDL.m
 - can be modified for any additional module configuration
- 1) Set global parameters
 - Set orbit parameters, module lifetime
 - Set design global parameters
 - pub_globals_bus_types.m
 - pub_globals_f6_elements.m
 - Load value global parameters
 - pub_value_globals.m
- 2a) Define scenario / perturbation model states
 - Simulation settings and state of uncertainty models
 - pub_Model_Cost_Parameters_default.m
- 2b) Define scenario / perturbation model parameters
 - Cost and value Levers
 - pub_Model_Enable_Selection_default.m
- 3) Select Configuration
 - Module and LV design
 - pub_baseline_config.m
- 4a) Run Model / save baseline results
 - pub_scenario_main.m
- 4b) Load new design for additional High Rate DL module and run model
 - Add new module to module configuration
 - Add single launch to launch configuration
 - Run model save results
- 5) Plot Data
 - plot_additional_HRDL.m



IBM JPL MIT

Massachusetts
Institute of
Technology

Georgia Tech



Sample Cases F6 (5yr) vs. Monolith (10yr)



- Run File – pub_f6_monolith_comparison.m
 - can be modified for any additional module configuration
- 1) Set global parameters
 - Set orbit parameters, module lifetime
 - Set design global parameters
 - pub_globals_bus_types.m
 - pub_globals_f6_elements.m
 - Load value global parameters
 - pub_value_globals.m
- 2a) Define scenario / perturbation model states
 - Simulation settings and state of uncertainty models
 - pub_Model_Cost_Parameters_default.m
- 2b) Define scenario / perturbation model parameters
 - Cost and value Levers
 - pub_Model_Enable_Selection_default.m
- 3) Select Configuration
 - Module and LV design
 - pub_baseline_5yr_config.m
- 4a) Run Model – (TLife Override version)
 - pub_scenario_main_TLife_override.m
- 4b) Load new design for monolith and repeat
 - Requires setting design horizon
 - Run model
- 5) Save / Plot Data
 - Plot_f6_monolith_comparison.m



IBM JPL MIT

Massachusetts
Institute of
Technology

Georgia Tech



Sample Cases

Altitude Trade

- Run File – trade_study_altitude.m
 - can be modified for any additional module configuration
- 1) Set global parameters
 - Set orbit parameters, module lifetime
 - Set design global parameters
 - pub_globals_bus_types.m
 - pub_globals_f6_elements.m
 - Load value global parameters
 - pub_value_globals.m
- 2a) Define scenario / perturbation model states
 - Simulation settings and state of uncertainty models
 - pub_Model_Cost_Parameters_default.m
- 2b) Define scenario / perturbation model parameters
 - Cost and value Levers
 - pub_Model_Enable_Selection_default.m
- 3) Select Configuration
 - Module and LV design
 - pub_baseline_config.m
- 4a) Run Model
 - pub_scenario_main.m
- 4b) Load new altitude (550) and repeat
 - Requires setting design horizon
 - Run model
- 4c) Load new altitude (600) and repeat
 - Requires setting design horizon
 - Run model
- 5) Save / Plot Data
 - plot_altitude_trade.m





PIVOT: Notes Regarding Initial Release



- Initial release of PIVOT used as a Programmer's toolkit for the modeled reference mission.
- Current Tool Limitations
 - Designs must currently adhere to a set structure of a single bus type
 - Tool accommodates maximum of six fractionated elements
 - Parameter modifications must be done manually
 - Following slides identify host files for design and model parameters
 - Value model only applicable to design reference mission
- Future work
 - GUI interface
 - plug-and-play capability
 - Design Characteristics
 - Additional Bus types
 - Additional F6 elements
 - Value Model



IBM JPL MIT

Massachusetts
Institute of
Technology

Georgia Tech



Design / Parameter Modifications



○ Design Modifications (Discrete Relations)

- Core module characteristics (`pub_globals_bus_types.m`)

- Subsystem mass and power budgets
- Reliability (FIT values)
- TRL Level (Cost-weighted average)
- Schedule parameters
 - Design time, long-lead items
- Module design constraints
 - Mass, power, propellant

- Fractionated components (`pub_globals_f6_elements.m`)

- Mass, power, reliability (FIT values), cost, and TRL for each element
- Component placement constraints – eliminate known invalid configurations
 - Used for volume, interface, power, or mass constraints
 - array of invalid configurations of F6 element combinations – column indicates element #
 - Ex. [0 1 0 0 0 1] indicates element 2 cannot be paired with element 6

- Launch Vehicle Selections (`pub_lv_input.m`)

- LV performance, cost, and reliability

○ Perturbation / scenario parameter selection

- Model selection (`pub_Model_Enable_Selection_default.m`)
- Model parameters (`pub_Model_Cost_Parameters_default.m`)



IBM JPL

Mit
Massachusetts
Institute of
Technology

Georgia Tech